

Space Acceleration Measurement Systems (SAMS)

System Specification Document

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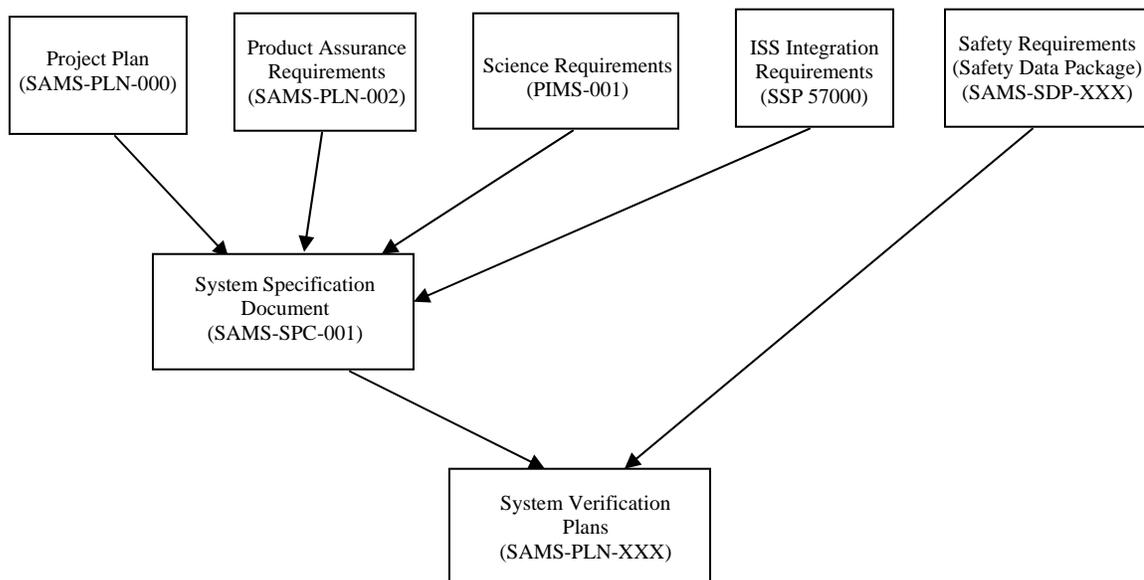
1.0 INTRODUCTION

The Space Acceleration Measurement Systems (SAMS) project will provide several microgravity acceleration measurement systems to support the NASA Office of Biological and Physical Sciences Research science experiments onboard the STS (Space Transportation System), ISS (International Space Station) and ELVs (Expandable Launch Vehicles). The microgravity acceleration measurement systems include vibratory sensor subsystems, angular acceleration subsystems and quasi-steady subsystems.

2.0 SCOPE

The purpose of this document is to establish the system requirements to be met by the SAMS project for the design, fabrication, verification, and operation of the microgravity accelerometer systems. All microgravity acceleration measurement systems that have been previously fabricated are not subject to the specifications in this document. For those items, refer to previous SAMS requirement documentation.

The following flow chart depicts the document flow for the design, fabrication, verification and operation of the SAMS systems.



3.0 APPLICABLE DOCUMENTS

The specifications have been derived from the following documents:

PIMS-001, Experiment Support Requirements Document for SAMS-II, September 25, 1995.

SAMS-PLN-000 Draft, Project Plan, October 2002.

SAMS-PLN-002, Product Assurance Plan, November 2002.

SSP 57000 Rev E, Pressurized Payloads Interface Requirements Document, November 1, 2000.

Information from the documents listed above that is not included in this document will not be formally verified. The specifications included in this document will be verified via the specific System Verification Plans.

4.0 REQUIREMENTS

Table I, below, describes each design requirement, its source, its source requirement, its rationale, its applicability (carrier that applies/subsystem that applies) and the verification number. For platform applicability, the choices are:

- ISS - International Space Station
- STS - Space Transportation System
- ELV - Expendable Launch Vehicle
- ESF – External Space Flight
- All - All of the above

For subsystem applicability, the choices are:

- VSS - Vibratory Sensor Subsystem (e.g. RTS, TSH-ES)
- AAS - Angular Acceleration Subsystem (e.g. Roll Rate Sensor)
- QSS - Quasi-Steady Subsystem (e.g. MAMS/OARE)
- CDHS - Control and Data Handling Subsystem (e.g. Software and associated hardware)
- GS - Ground Subsystem (e.g. TSC hardware and software)
- All - All of the above (excludes GSES)
- GSES - Ground Support Equipment Subsystem (Checkout and shipping hardware)

The tracking numbers are assigned for ease of tracking and referencing each requirement. These numbers are grouped by source documents, which are indicated by the first number in the tracking number. The following details the numbering scheme:

- 0xxx – PIMS-001, Experiment Support Requirement Document for SAMS-II
- 1xxx – SAMS-PLN-000, Project Plan
- 2xxx – SAMS-PLN-002, Product Assurance Plan
- 3xxx – SSP 57000, Pressurized Payload Interface Requirements Document

The verification numbers are assigned through the verification plans. An explanation of the numbering scheme is described in each verification plan. For consistency the verification number is tied to a requirement; therefore, the different verification plans' verification numbers have the same last three digits for a common requirement. The middle three digits represented by “xxx” are the document number of the verification plan. Using this convention, the VRDSs can be tracked by the verification plan and the requirement for which they are applicable. The verification numbers with “tbd” have not been assigned a number because an applicable verification plan has not been created. These numbers will be assigned in the future. The verification number for SSP 57000 requirements are listed on the table in Appendix A.

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
0001	VRD-xxx-001	PIMS-001, Table 2, 1.	Acquire microgravity acceleration data.	Acquire microgravity acceleration data.	When performing microgravity experiments, disturbances caused by various inputs affect the experiments' results. Thus the acceleration data must be collected so it can be used to help understand the results.	All*/All
0002	VRD-xxx-100	PIMS-001, Table 2, 1.a	Measure accelerations with an accuracy and resolution better than the acceleration environment envelope of the ISS program.	The system shall provide measurements in three axes of the microgravity acceleration where the noise level for each axis is 10dB below the curve defined by: 1.8 µgrms for data from 0.01 Hz to 0.1 Hz, 18 µgrms x frequency for data from 0.1 to 100 Hz, 1.8 mgrms for data from 100Hz to 300 Hz.	For experiments on ISS, a specification for the required environment was defined. The acceleration systems need to be able to resolve the specified environment.	All*/VSS
0003	VRD-xxx-002	PIMS-001, Table 2, 1.a	Measure accelerations with an accuracy and resolution better than the acceleration environment envelope of the ISS program.	The system shall provide acceleration measurements in which the maximum error is less than 10% of the measured value for measurements that exceed the system noise floor requirement (reference 3.1.1) across the frequency range of 0.01Hz to 300 Hz.	For experiments on ISS, a specification for the required environment was defined. The acceleration systems need to be able to resolve the specified environment.	All*/VSS
0004	VRD-xxx-101	PIMS-001, Table 2, 1.a	Measure accelerations with an accuracy and resolution better than the acceleration environment envelope of the ISS program.	The magnitude of the system's frequency response shall be within 1 dB of/or 1 (output/input) over the frequency range of 0.01 Hz to 300 Hz.	For experiments on ISS, a specification for the required environment was defined. The acceleration systems need to be able to resolve the specified environment.	All*/VSS
0005	VRD-xxx-tbd	PIMS-001, Table 2, 1.a	Measure accelerations with an accuracy and resolution better than the acceleration environment envelope of the ISS program.	The magnitude of the system's frequency response shall be within 1 dB of/or 1 (output/input) over the frequency range of 0.00018 Hz to 0.01 Hz.	For experiments on ISS, a specification for the required environment was defined. The acceleration systems need to be able to resolve the specified environment.	All*/QSS
0006	VRD-xxx-200	PIMS-001, Table 2, 1.a	Measure accelerations with an accuracy and resolution better than the acceleration environment envelope of the ISS program.	The system shall produce measurement data from accelerometers with measurement axes mutually orthogonal within plus or minus 0.1 degree.	For experiments on ISS, a specification for the required environment was defined. The acceleration systems need to be able to resolve the specified environment.	All*/VSS, AAS
0007	VRD-xxx-300	PIMS-001, Table 2, 1.a	Measure accelerations with an accuracy and resolution better than the acceleration environment envelope of the ISS program.	SAMS shall produce a matrix for each sensor system that includes all software coefficients and location coordinates.	Software coefficients and location coordinates are necessary for the accuracy of the data. This matrix will insure all values are recorded and filed for each sensor system.	All*/VSS, AAS, QSS
0008	VRD-xxx-301	PIMS-001, Table 2, 1.b	Acquire the acceleration data with correlated time information	SAMS shall correlate time by acquiring time from the command and data handling subsystem host rack (or central timer) and syncing all sensors to it. Provisions will be made to sync with sensor host rack (or central timer) when rack to rack communications are down.	Experiment data needs to be correlated with the acceleration data. It is proposed that it be done by correlating to ISS time.	All*/VSS, AAS, QSS, CDHS

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
0009	VRD-xxx-003	PIMS-001, Table 2, 1.c	Measure acceleration within selectable frequency range	The following selectable frequency ranges shall be provided: 3 Hz \pm 1 Hz 6 Hz \pm 2 Hz 12 Hz \pm 2 Hz 15 Hz \pm 2 Hz 25 Hz \pm 10 Hz 50 Hz \pm 25 Hz 100 Hz \pm 50 Hz 200 Hz \pm 75 Hz 300 Hz \pm 100 Hz	Selectable frequency ranges are desired. The requirement is written to not restrict the user by the use of filters. PIMS has defined these frequencies to be comparable to SE frequencies and meet processing needs.	All*/VSS
0010	VRD-xxx-004	PIMS-001, Table 2, 1.d	Measure acceleration in, on and/or near the experiment sample/ chamber/ apparatus.	As small as technology and cost allows. An agreement of the location shall be made with the customer.	Sensor subsystem must be small in order to be located in the area of interest , but it is realized that technology, cost and schedule limit the ability to minimize the size. Design needs to show options.	All*/VSS, AAS, QSS
0011	VRD-xxx-302	PIMS-001, Table 2, 2.	Allocate control of SAMS-II.	Allocate control of acceleration systems/subsystems.	In order to minimize operational requirements on the project team, it is desired to have other parties such as the cadre, science team, or other user be able to adjust the parameters on the sensor they are using.	All/CDHS
0012	VRD-xxx-005	PIMS-001, Table 2, 2.	Allocate control of SAMS-II.	Shall provide external triggers.	To allow SAMS to initiate experiment operations. Functionality to be incorporated with first customer request and funding availability.	All/CDHS
0013	VRD-xxx-302	PIMS-001, Table 2, 2.a	PI control of parameters.	The command and data handling subsystem or the sensors shall have the ability to allow PI users to command with the proper security/safety measures.	The desire is to have the PI control the parameters from their remote site, limiting SAMS involvement. This is a maturing requirement based on customer requests and funding. Note: For ISS, due to restrictions a PI will command systems through their own payload.	All/CDHS
0014	VRD-xxx-303	PIMS-001, Table 2, 2.b	On-orbit crew control of parameters.	The command and data handling subsystem or the sensors shall be able to accept commands directly from the crew laptops with the proper security/safety measures.	The desire is to have the crew control the parameters on board, limiting SAMS involvement.	ISS, STS/CDHS
0015	VRD-xxx-304	PIMS-001, Table 2, 3.	Supply acceleration information to users.	Supply acceleration information to users.	Supply data to users in near real time and archive data for future use by the users.	All/All
0016	VRD-xxx-305	PIMS-001, Table 2, 3.a	Supply information in a selectable format.	Supply information in a selectable format.	This refers to the near real time data. There should be standard data displays.	As required/CDHS, GS
0017	VRD-xxx-306	PIMS-001, Table 2, 3.b	Supply information within a selectable amount of time.	Supply information within a selectable amount of time.	User needs to be able to display archived data.	All/CDHS, GS

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
1001	VRD-xxx-tbd	SAMS-PLN-000, Section 2.2	SAMS shall have systems that are available 90% of the time when there is microgravity research being performed.	A system shall be developed so an availability of 90% can be achieved.	Success criteria in project plan.	All/All
1002	VRD-xxx-307	SAMS-PLN-000, Section 2.2	SAMS shall have systems that are available 90% of the time when there is microgravity research being performed.	Data flow shall minimize single point failures.	This is to limit adverse affects of non functioning carrier/facility systems. Since the data flow paths can be interrupted on ISS, multiple data paths should be utilized when practical.	ISS/All
1003	VRD-xxx-tbd	SAMS-PLN-000, Section 2.2	SAMS shall have systems that are available 90% of the time when there is microgravity research being performed.	Ground software shall be checked with flight software using a carrier/facility simulator.	This will reduce troubleshooting during the mission.	ISS/All, GSES
1004	VRD-xxx-308	SAMS-PLN-000, Section 2.2	SAMS shall have systems that are available 90% of the time when there is microgravity research being performed.	All uploads shall be verifiable on orbit prior to implementing on orbit.	Before executing anything on orbit, the upload needs to be verified to reduce down time.	All/All
1005	VRD-xxx-006	SAMS-PLN-000, Section 6.2, 1	All systems shall follow the SAMS System Philosophy shown in figure 6.2-1.	All systems shall follow the SAMS System Philosophy shown in figure 6.2-1.	A consistent approach is required to maximize the usage of every developed item. This is especially important in long lead items such as software.	All/All, GSES
1006	VRD-xxx-007	SAMS-PLN-000, Section 6.2, 2	Systems shall be designed to create the least impact on the supported experiment or facility.	Any new developed system shall have less volume, less mass, less power than the system it is replacing.	The SAMS System is a support instrument to measure one parameter of the research experiment. Therefore, the system needs to minimally use resources. This requirement will be applied to development of new systems compared to existing systems.	All/All
1007	VRD-xxx-309	SAMS-PLN-000, Section 6.3, 1	For remote operations, the ground crew shall be able to easily access system health and status, troubleshooting data, and error information.	All commands shall have acknowledgements sent to SAMS remote operations that will tell who sent the command, when was it sent, what it was, and was it executed. In addition, all processes will notify the ground when they start and stop or completed a task.	Several users may be sending commands to the system. The acknowledgements will allow the SAMS team to verify the system activity is nominal. This is a lesson learned.	All/CDHS
1008	VRD-xxx-309	SAMS-PLN-000, Section 6.3, 1	For remote operations, the ground crew shall be able to easily access system health and status, troubleshooting data, and error information.	Temperature readings shall be downlinked. When output by the system hardware, voltages and currents shall be downlinked.	In order to minimize operations and save costs, the health and status of the remote systems need to be accessed quickly. This is especially important when performing off-nominal trouble shooting. Previous experience shows that data such as temperatures, voltages, status readings, unique packet identifiers for data types and replay data, variable indication, diagnostic messages from processes, and incrementing indicator in the packet are required to minimize troubleshooting time.	All/CDHS
1009	VRD-xxx-309	SAMS-PLN-000, Section 6.3, 1	For remote operations, the ground crew shall be able to easily access system health and status, troubleshooting data, and error information.	Unique packet identifiers shall be assigned for each different data structure.	To identify different data structure.	All/All

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1010	VRD-xxx-tbd	SAMS-PLN-000, Section 6.3, 1	For remote operations, the ground crew shall be able to easily access system health and status, troubleshooting data, and error information.	For replay data, a unique packet identifier shall be assigned which is different from the realtime data.	To identify replay data from realtime data.	All/All
1011	VRD-xxx-309	SAMS-PLN-000, Section 6.3, 1	For remote operations, the ground crew shall be able to easily access system health and status, troubleshooting data, and error information.	The ability to downlink diagnostic data shall be incorporated. In addition, a subset of data shall be placed in housekeeping health data.	Current status is always known.	All/All
1012	VRD-xxx-310	SAMS-PLN-000, Section 6.3, 1	For remote operations, the ground crew shall be able to easily access system health and status, troubleshooting data, and error information.	An incrementing indicator shall be given to every packet.	This is to verify that the data is updating	All/All
1013	VRD-xxx-311	SAMS-PLN-000, Section 6.3, 1	For remote operations, the ground crew shall be able to easily access system health and status, troubleshooting data, and error information.	System shall have file downlink capability.	Additional troubleshooting tool. This is a lesson learned.	ISS/All
1014	VRD-xxx-312	SAMS-PLN-000, Section 6.3, 2	The system software shall have the capability to be updated by uplinking new system software.	The system software shall have the capability to be updated by uplinking new system software.	SAMS will be able to make marginal upgrades without any up mass. This will allow for more customer focus for each mission and minimize trying to develop a system software that does it all from the beginning.	ISS/All
2001	VRD-xxx-400	SAMS-PLN-002, Section 1.4	SAMS shall issue assurance status reports of deliverables at Pre-Ship reviews.	Issue assurance status reports of deliverables at Pre-Ship reviews.		All/All
2002		Deleted Requirement				
2003	VRD-xxx-201	SAMS-PLN-002, Section 3.4.1	All structural elements including associated interfaces, fasteners, and welds in the primary load path are considered safety critical and shall be analyzed per the guidelines of SSP 52005.	Safety critical hardware shall be analyzed per the guidelines of SSP 52005.	The extensive verifications required for Safety-Critical Structures (SCS) are a due to the critical flight safety concern of the structural integrity of the Space Transportation System (STS) and International Space Station (ISS) payloads.	All/VSS, AAS, QSS, CDHS
2004	VRD-xxx-202	SAMS-PLN-002, Section 3.4.5	A Fracture Control Program shall be implemented to meet the requirements of NASA-STD-5003 and SSP 30558.	Implement a Fracture Control Program (minimally Plan and Summary) to meet the requirements of NASA-STD-5003 and SSP 30558.	To ensure components whose failure would represent a catastrophic hazard are safe from failure throughout the mission.	All/VSS, AAS, QSS, CDHS
2005	VRD-xxx-201	SAMS-PLN-002, Section 3.4.7	The SAMS Project will verify structural integrity by analysis alone without strength testing using the criteria and design factors specified in SSP 52005. Criteria and rationale shall be reviewed and approved by the GRC Verification Readiness Review Panel and the Payload Carrier.	Design shall encompass load factors and random vibration environments for various carrier locations. Maximum allowable quasi-static and random vibration loads will be determined for the sensor subsystem.	Maximize locations where the sensor subsystems can be mounted.	All/VSS, AAS, QSS, CDHS

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2006	VRD-xxx-203	SAMS-PLN-002, Section 3.4.8	Qualification random vibration tests shall be performed on dedicated test hardware. NASA-STD-7001 will be used in conjunction with carrier mission requirements to define the random vibration environment.	Perform qualification test on dedicated test hardware (engineering/prototype units), produced from the same drawings and using the same materials, tooling, manufacturing processes, inspection methods, and level of personnel competency as used for the flight hardware.	Qualification tests demonstrate, with margin, the design adequacy of the hardware for its intended mission use.	All/VSS, AAS, QSS, CDHS
2007	VRD-xxx-204	SAMS-PLN-002, Section 3.4.8	Acceptance or workmanship random vibration tests shall be performed on flight system. NASA-STD-7001 will be used in conjunction with carrier mission requirements to define the random vibration environment.	Perform an acceptance or workmanship random vibration test on all units designated for flight.	To demonstrate satisfactory performance relative to the expected environment and to reveal inadequacies in workmanship and material integrity.	All/VSS, AAS, QSS, CDHS
2008	VRD-xxx-205	SAMS-PLN-002, Section 3.4.8	Low-Level sine-sweep testing shall be performed on engineering hardware.	Low-Level sine-sweep testing shall be performed on prototype hardware.	Low-Level sine-sweep tests performed to determine natural frequencies and provide finite element model validation.	All/VSS, AAS, QSS, CDHS
2009	VRD-xxx-206	SAMS-PLN-002, Section 3.4.10	A kinematics analysis shall be performed on all experiment mechanical operations.	Perform kinematics analysis.	To show that each mechanism can perform satisfactorily and has adequate design margin under worst-case conditions; satisfactory mechanical component clearances exist for stowed, operational configuration and mechanical operation and all mechanical elements are capable of withstanding the worst-case loads that may be encountered.	All/VSS, AAS, QSS, CDHS
2010	VRD-xxx-206	SAMS-PLN-002, Section 3.4.10	Qualification testing for each mechanical operation shall only be performed for integration in which the hardware is subjected to thermal gradients greater than 60°.	Perform qualification testing for each mechanical operation for integration in which the hardware is subjected to thermal gradients greater than 60°.	To show that each mechanism can perform satisfactorily and has adequate design margin under worst-case conditions; satisfactory mechanical component clearances exist for stowed, operational configuration and mechanical operation and all mechanical elements are capable of withstanding the worst-case loads that may be encountered.	All/VSS, AAS, QSS, CDHS
2011	VRD-xxx-206	SAMS-PLN-002, Section 3.4.10	Acceptance testing shall be performed to demonstrate that the installation of each mechanical device is correct and that no problems exist that will prevent the proper operation of the mechanism during mission life.	Perform verification testing.	To show that each mechanism can perform satisfactorily and has adequate design margin under worst-case conditions; satisfactory mechanical component clearances exist for stowed, operational configuration and mechanical operation and all mechanical elements are capable of withstanding the worst-case loads that may be encountered.	All/VSS, AAS, QSS, CDHS
2012	VRD-xxx-207	SAMS-PLN-002, Section 3.4.11	Verification of payload structures containing trapped volumes shall be by analysis.	Design & analyze payload structures containing trapped volumes per NSTS 1700.7 and ISS Addendum, NSTS 18798, and SSP 52005 when exposed to the worst-case depressurization/re-pressurization environments based on carrier location.	Verification that payload structure is designed to maintain positive margins of safety when exposed to the worst-case depressurization/re-pressurization environments based on carrier location.	All/VSS, AAS, QSS, CDHS

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2013	VRD-xxx-208	SAMS-PLN-002, Section 3.4.12	The SAMS Project shall develop a Fastener Control Plan as a Phase I safety submittal.	Develop a Fastener Control Plan.	Maintain fastener integrity.	All/VSS, AAS, QSS, CDHS
2014	VRD-xxx-tbd	SAMS-PLN-002, Section 3.4.14	All Ground Support Equipment (GSE) shall meet the facility requirements where it will operate.	Perform verification testing/analyses on GSE.	This requirement identifies the need to meet safety and integration requirements at applicable facilities.	All/GSES
2015	VRD-xxx-102	SAMS-PLN-002, Section 3.5	EMC verifications shall be demonstrated via test to the levels required by the Carrier or customer.	EMC verifications shall be demonstrated via test to the levels required by the carrier or customer.	To ensure the system will not generate electromagnetic interference that interferes with its own missing objectives or the operation and safety of concurrently operating systems. Secondly, to ensure the system can operate in the mission's defined electromagnetic environment.	All/VSS, AAS, QSS, CDHS
2016	VRD-xxx-209	SAMS-PLN-002, Section 3.6	Vacuum (if applicable per carrier), thermal and humidity test shall demonstrate the system under test will: a. Properly operate in the flight environment, specifically vacuum, temperature, and humidity. b. Properly control the thermal environment of temperature sensitive items or levy that requirement on an integrator. c. Survive the temperature and humidity conditions of transportation (e.g. truck, plane, space vehicle), storage and pre and post conditions of the carrier.	Perform vacuum (if applicable per carrier), thermal and humidity tests.	Tests shall demonstrate the system under test will: a.) Properly operate in the flight environment, specifically vacuum, temperature, and humidity. b.) Properly control the thermal environment of temperature sensitive items or levy that requirement on an integrator. c.) Survive the temperature and humidity conditions of transportation (e.g. truck, plane, space vehicle), storage and pre and post conditions of the carrier. These tests will also act as an environmental screening to stimulate latent defects to maximize infant mortality failures.	All/VSS, AAS, QSS, CDHS
2017	VRD-xxx-210	SAMS-PLN-002, Section 3.6.4	The SAMS project shall perform Thermal Analysis on all systems.	Perform Thermal Analysis on all systems. The analysis shall identify the following: a.) Heat sources and magnitude b.) Methods employed to dissipate the heat from the sources. c.) List of operating temperature ranges of the components. d.) Environmental conditions and design criteria. e.) An assessment of the thermal design and identification of additional analysis needed. f.) An evaluation of the susceptibility to humidity extremes. g.) Identification of any special testing requirements or conditions.		All/VSS, AAS, QSS, CDHS
2018	VRD-xxx-103	SAMS-PLN-002, Section 3.7.1	A minimum of one hundred hours of failure-free operation shall be accumulated for SAMS components, subsystems or systems prior to hardware turnover.	Verify a minimum of one hundred hours of failure-free operation is accumulated on SAMS components, subsystems or systems prior to hardware turnover. Life of each component/subsystem shall be checked that there is enough life to perform mission.	For systems under test that contain electronic parts, a burn-in is required to stimulate infant mortality failures.	All/VSS, AAS, QSS, CDHS

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
2019	VRD-xxx-313	SAMS-PLN-002, Section 3.7.2	SAMS shall perform a mission simulation test.	A minimum continuous four-day operational test shall be conducted which exercises all equipment and software commands for missions exceeding four days.		ISS/All
2020	VRD-xxx-tbd	SAMS-PLN-002, Section 3.7.2	SAMS shall perform a mission simulation test.	For other missions, the mission simulation test shall be defined by the duration and operation constraints of the mission.		STS, ELV, ESF/All
2021	VRD-xxx-313	SAMS-PLN-002, Section 3.7.3	SAMS shall perform End-to-End compatibility tests per customer requirements.	Perform End-to-End compatibility tests per customer requirements.	This may require simulation of the carrier/facility interface. This is meant to test data flow from initiation to user delivery. Test may be performed in parts.	All/All
2022	VRD-xxx-401	SAMS-PLN-002, Section 4.1	SAMS shall conduct appropriate safety analyses both at the system and subsystem levels.	Perform a flight safety analysis per NSTS 1700.7, NSTS 1700.7 ISS Addendum and NSTS/ISS 13830C.		All/VSS, AAS, QSS, CDHS
2023	VRD-xxx-104	SAMS-PLN-002, Section 5.2.5	All new flight hardware shall be tested for single-event effects.	Test all new flight hardware for single-event effects radiation.		All/VSS, AAS, QSS, CDHS
2024	VRD-xxx-105	SAMS-PLN-002, Section 5.2.8	EEE parts shall be inspected prior to their assembly into flight systems or subsystems to ensure they are free of any debris, defects or other manufacturing faults that would interfere with their form, fit, and function.	EEE parts shall be inspected prior to their assembly into flight systems or subsystems.	To ensure they are free of any debris, defects or other manufacturing faults that would interfere with their form, fit, and function.	All/VSS, AAS, QSS, CDHS
2025	VRD-xxx-106	SAMS-PLN-002, Section 5.2.8	EEE parts that are safety or mission critical shall be tested to verify and certify their electrical performance and protect against counterfeit or noncompliant parts prior to assembly into flight systems or subsystems.	Test components made with EEE parts that are safety or mission critical. To verify and certify their electrical performance and protect against counterfeit or noncompliant parts prior to assembly into flight systems or subsystems.	To verify and certify their electrical performance and protect against counterfeit or noncompliant parts prior to assembly into flight systems or subsystems.	All/VSS, AAS, QSS, CDHS
2026	VRD-xxx-211	SAMS-PLN-002, Section 5.3.2	All mechanical parts shall be inspected prior to their assembly into flight systems or subsystems to ensure they are free of any debris, defect or other material or substance that would interfere with their function.	Inspect all mechanical parts prior to their assembly into flight systems or subsystems.	To ensure they are free of any debris, defect or other material or substance that would interfere with their function.	All/VSS, AAS, QSS, CDHS
2027	VRD-xxx-211	SAMS-PLN-002, Section 5.3.2	All mechanical parts that provide rotational, translational or other movements shall be tested for full range of motion, and inspected for freedom of motion (resistance) if inaccessible after assembly.	Test all mechanical parts that provide rotational, translational or other movements.	To ensure full range of motion and freedom of motion (resistance) .	All/VSS, AAS, QSS, CDHS
2028		Deleted Requirement				

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
2029	VRD-xxx-212	SAMS-PLN-002, Section 6.1	SAMS shall compile a Material Identification Usage List (MIUL) during the design process.	The MIUL will identify individual material applications specified in design drawings; their respective pertinent material ratings; and other pertinent material application features essential for evaluation of the material application.	To provide the framework to assure proper selection, application, processing, inspection, and testing of the chosen materials for the project subsystems and equipment.	All/VSS, AAS, QSS, CDHS
2030	VRD-xxx-212	SAMS-PLN-002, Section 6.1	SAMS shall prepare MUAs for materials that are not A-rated.	Prepare rationale and gain approval for non-A-rated materials usage.	Ensure the materials selected for space flight hardware meet safety requirements.	All/VSS, AAS, QSS, CDHS
2031	VRD-xxx-220	SAMS-PLN-002, Section 6.4.h	Closeout photographs shall be taken during assembly to show materials and configurations hidden after final assembly.	Prior to final assembly, closeout photographs shall be taken of the hardware showing materials and configurations of internal hardware.	Document areas of assembly no longer able to be visually inspected.	All/VSS, AAS, QSS, CDHS
2032	VRD-xxx-tbd	SAMS-PLN-002, Section 6.4.1	All organic materials used in the pressurized environment on the International Space Station shall be evaluated for fungus resistance prior to selection.	SAMS does not use organic materials.	Since SAMS does not use organic materials, this requirement will not be part of the verification plans.	None
2033	VRD-xxx-214	SAMS-PLN-002, Section 6.4.1.3	In the event that a SAMS system is used in exterior applications, the system shall meet the out-gassing requirements of JSC SP-R-0022A that establishes the out-gassing requirements and test guidelines for materials used in the space thermal vacuum environment.	SAMS shall meet the out-gassing requirements of JSC SP-R-0022A if the hardware is to be used in exterior applications.		ESF/VSS, AAS, QSS, CDHS
2034	VRD-xxx-tbd	SAMS-PLN-002, Section 6.4.1.5	If the fluid is other than LOX or GOX, the project shall develop a fluid compatibility test plan and submit it to the GRC M&P engineer prior to testing. When a material fails the criteria of either NASA-STD-6001 tests or project specific test plan, the project manager with GRC M&P concurrence shall determine how to proceed.	SAMS does not use fluid systems.	Since SAMS does not use fluid systems, this requirement will not be part of the verification plans.	None
2035	VRD-xxx-215	SAMS-PLN-002, Section 6.4.1.6	SAMS will design systems such that all materials used in habitable flight compartments shall meet the off-gassing requirement of NASA-STD-6001 Test 7 or by analysis.	SAMS shall meet the off-gassing requirement of NASA-STD-6001 Test 7 or by analysis.	Excessive off-gassing can cause a health issue for the crew.	All/VSS, AAS, QSS, CDHS
2036	VRD-xxx-216	SAMS-PLN-002, Section 6.4.1.7	The SAMS project shall perform a Fungus Assessment on all International Space Station developed elements or payloads per SSP 30233.	A Fungus Assessment shall be performed on all ISS hardware.	Fungus can cause a health issue for the crew.	ISS/VSS, AAS, QSS, CDHS

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
2037	VRD-xxx-217	SAMS-PLN-002, Section 6.4.1.8	The SAMS project shall perform an Atomic Oxygen (AO) assessment on all elements or payloads exposed to the Low Earth Orbit (LEO) environment.	Perform an Atomic Oxygen (AO) assessment on all elements or payloads exposed to the LEO environment.	The AO assessment will indicate that the performance requirements are not compromised in the LEO environment.	ESF/VSS, AAS, QSS, CDHS
2038	VRD-xxx-402	SAMS-PLN-002, Section 7.2.1	SAMS subsystems shall have a reliability of 90% with a mission duration and duty cycle appropriate for the specific hardware.	Design subsystem to provide an MTBF in excess of the mission duration, considering duty cycle, with a 90% confidence level.	Provide confidence of mission success balanced against cost and availability of parts.	All/VSS, AAS, QSS, CDHS
2038.1	VRD-xxx-402	SAMS-PLN-002, Section 7.2.1	SAMS subsystems shall have a reliability of 90% with a mission duration and duty cycle appropriate for the specific hardware.	Vibratory sensor subsystems shall be designed for a 12 year life with a 50% duty cycle.	Values chosen to support availability analysis.	All/VSS
2038.2	VRD-xxx-tbd	SAMS-PLN-002, Section 7.2.1	SAMS subsystems shall have a reliability of 90% with a mission duration and duty cycle appropriate for the specific hardware.	Quasi-steady subsystems shall be designed for a 10 year safety life and 5 year functional life with a 90% duty cycle.	Values chosen to support availability analysis.	All/QSS
2038.3	VRD-xxx-tbd	SAMS-PLN-002, Section 7.2.1	SAMS subsystems shall have a reliability of 90% with a mission duration and duty cycle appropriate for the specific hardware.	Angular acceleration subsystems shall be designed for a 12 year safety life and 10 year functional life with a 90% duty cycle.	Values chosen to support availability analysis.	All/AAS
2038.4	VRD-xxx-tbd	SAMS-PLN-002, Section 7.2.1	SAMS subsystems shall have a reliability of 90% with a mission duration and duty cycle appropriate for the specific hardware.	Control and data handling subsystem shall be designed for a 10 year safety life and 5 year functional life with a 90% duty cycle.	Values chosen to support availability analysis.	All/CDHS
2039	VRD-xxx-314	SAMS-PLN-002, Section 7.2.2	The SAMS project shall monitor operating times of all flight subsystems and relate it to the overall equipment life.	Provide a summation of operating life on components.	Document and review operating time on flight subsystems.	All/All
2040	VRD-xxx-403	SAMS-PLN-002, Section 7.2.3	The SAMS project shall perform a Failure Modes and Effects Analysis (FMEA) on select systems or subsystems based on initial reliability analysis and criticality to mission success and safety. Whenever an FMEA is performed, a Critical Items List (CIL) shall be compiled to document the safety critical and mission critical failure modes.	Perform a FMEA and compile a CIL per SSP 30234.	Analyze and document failure modes and assess criticality.	All/All
2041	VRD-xxx-404	SAMS-PLN-002, Section 7.2.4	The project shall provide a list of single point failures that would lead to loss of life, vehicle, or critical mission support capability.	Compile a list of single point failures that would lead to loss of life, vehicle, or critical mission support capability.	Analyze and document single point failure scenarios.	All/All
2042	VRD-xxx-405	SAMS-PLN-002, Section 7.2.5	SAMS shall provide a description of the automatic or manual failure detection, isolation and recovery methods supported by the design under analysis.	Report the automatic or manual failure detection, isolation and recovery methods supported by the design under analysis.	Document the automatic or manual failure detection, isolation and recovery methods supported by the design under analysis.	All/VSS, AAS, QSS, CDHS

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
2043	VRD-xxx-406	SAMS-PLN-002, Section 7.2.6	A fault tree analysis shall be performed on flight hardware per carrier.	Perform a fault tree analysis per SSP 30309. (Reference GRC-W0510.060 for a guideline in constructing the fault tree.)	Requirement per SSP 30309 determines all events and combination of events that can lead to an undesired event.	ISS/All
2044	VRD-xxx-218	SAMS-PLN-002, Section 7.2.7.1	Thermographic mapping or thermal analysis shall be performed on safety critical components (per CIL).	Perform thermographic mapping or thermal analysis on safety critical components (per CIL) to use in the de-rating of the component.	Determine actual thermal operating environment for safety critical components.	All/VSS, AAS, QSS, CDHS
2045	VRD-xxx-107	SAMS-PLN-002, Section 7.2.8	Worst-case analysis shall be performed by the project on all circuits that support safety and mission critical functions.	Perform worst-case analysis on all circuits that support safety and mission critical functions.	Analyze circuits that support safety and mission critical functions by the most conservative method.	All/VSS, AAS, QSS, CDHS
2046	VRD-xxx-407	SAMS-PLN-002, Section 7.2.9	Performance trending shall be performed by the SAMS project on bias shift and scale factor, which is measured during calibration of the sensor head.	Perform performance trending on bias shift and scale factor, which is measured during calibration of the sensor head.	Document a project performance indicator.	All/VSS, AAS, QSS
2047	VRD-xxx-tbd	SAMS-PLN-002, Section 7.3.1	SAMS shall have an availability of 90% for ISS missions.	Perform availability analysis	Document that availability is greater than 90%	ISS/All
2048	VRD-xxx-408	SAMS-PLN-002, Section 7.4.1	Based upon the system availability requirement, MTTR (Mean Time To Repair) allocations to the component/ORU/LRU level shall be developed.	Develop MTTR allocations to the component/ORU/LRU level .		All/VSS, AAS, QSS, CDHS
2049	VRD-xxx-408	SAMS-PLN-002, Section 7.4.2	Limited-life items shall be identified and managed by means of a Limited-Life Plan.	Compile a Limited-Life Plan, which will be submitted to the NASA Project Manager for acceptance.	Document the procedure for handling limited life items.	All/All
2050	VRD-xxx-408	SAMS-PLN-002, Section 7.4.3	When there are anticipated maintenance activities during the mission, the project shall perform a qualitative maintainability analysis.	Perform qualitative maintainability analysis when maintenance activities are anticipated during the mission.	When indicated, quantitatively analyze maintainability.	All/All
2051	VRD-xxx-408	SAMS-PLN-002, Section 7.4.4	Provisions for failure detection, isolation, annunciation, and recovery processes may include manual control of the automated processes and shall provide the following: a. Capability to detect and isolate failures that could manifest a catastrophic or critical hazard b. Confirmation of a restored function	When manual control of the automated processes is included, provide the following: a. Capability to detect and isolate failures that could manifest a catastrophic or critical hazard b. Confirmation of a restored function	Provisions for when manual control is used to control automated processes.	All/VSS, AAS, QSS, CDHS
2052	VRD-xxx-408	SAMS-PLN-002, Section 7.4.5	When there are anticipated maintenance activities during the mission, the project shall prepare a Maintainability Assessment Report.	Prepare a Maintainability Assessment Report	Meet the maintainability requirement	All/All

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
2053	VRD-xxx-408	SAMS-PLN-002, Section 7.4.6	When there are anticipated maintenance activities during the mission, the project shall conduct an analysis to determine which items require a preventive maintenance schedule that will minimize the amount of system down time.	Conduct an analysis to determine which items require a preventive maintenance schedule that will minimize the amount of system down time.	Meet the maintainability requirement	All/All
2054	VRD-xxx-219	SAMS-PLN-002, Section 8.7.1	This assures that before approving the unit for shipment, all components to be turned over shall meet the surface level cleanliness requirements of the particular project. Quality Assurance or project representative, as required, shall inspect and certify that the unit(s) being shipped meet the requirements.	Certify that all units being shipped meet the surface level cleanliness requirements stipulated by the carrier or customer. Certification shall be per quality assurance or project representative, as required.	Verification that units shipped are externally clean.	All/VSS, AAS, QSS, CDHS
2055	VRD-xxx-409	SAMS-PLN-002, Section 8.10	The project shall review NASA Parts Advisories and GIDEP Failure Experience reports according to the requirements of NPG 8735.1 and respond to project review teams as to the applicability to project systems, location of affected system, criticality identification from the FMEA/CIL and disposition for design reviews.	Perform GIDEP Check for design reviews, verification readiness reviews and pre-ship reviews.	Increase system reliability by identifying components that have known defects.	All/All
2056	VRD-xxx-220	SAMS-PLN-002, Section 8.14	Formal verification and sign-off that flight as-built systems conform to as-designed documentation shall be the responsibility of the project.	Verify that flight as-built systems conform to as-designed documentation.		All/All
2057	VRD-xxx-410	SAMS-PLN-002, Section 8.15	Prior to the Pre-ship Review the SAMS project shall assure that all deliverable end-items including the Acceptance Data Package are in accordance with program/project requirements.	Provide all deliverable end-items, including the Acceptance Data Package, in accordance with program/project requirements.	Meets contractual obligations, as well as, ensures a quality product is delivered.	All/All
2058	(Deleted)	(Deleted)	(Deleted)	(Deleted)	(Deleted)	(Deleted)
2059	VRD-xxx-315	SAMS-PLN-002 section 10.1.2	SAMS shall develop the software documentation shown in the table in section 10.1.2 of the PAP for medium and low control level SAMS software efforts.	SAMS shall develop the software documentation shown in the table in section 10.1.2 of the PAP for medium and low control level SAMS software efforts.	These products help ensure a better quality software product.	All/All
2060	VRD-xxx-316	SAMS-PLN-002 section 10.3	All SAMS software controlled items shall have hardware controls in the event of a hazardous condition.	All SAMS software controlled items shall have hardware controls in the event of a hazardous condition.	Software is zero fault tolerant, need hardware backup.	All/All
2061	VRD-xxx-317	SAMS-PLN-002 section 10.4	SAMS software shall undergo a minimum of 100 continuous hours of failure-free burn-in testing.	SAMS software shall undergo a minimum of 100 continuous hours of failure-free burn-in testing.	Burn-in tests help to ensure a reliable product.	All/All

Tracking #	Verification #	Source Document #, Location	Source Requirement	Design Requirement	Rationale	Applicability (Platform/Subsystem)
2062	VRD-xxx-317	SAMS-PLN-002 section 10.4	SAMS software shall undergo a detailed functional acceptance test that will test both nominal and boundary conditions.	SAMS software shall undergo a detailed functional acceptance test that will test both nominal and boundary conditions.	Functional acceptance tests not only verify and validate the software, but helps to ensure a reliable product.	All/All
2063	VRD-xxx-317	SAMS-PLN-002 section 10.7.3	SAMS software shall undergo a functional acceptance test prior to release. This test shall verify each capability of the software and ensure all software performance requirements have been satisfied as well as testing non-nominal and boundary conditions. Results of tests will be documented in a test report.	SAMS software shall undergo a functional acceptance test prior to release. This test shall verify each capability of the software and ensure all software performance requirements have been satisfied as well as testing non-nominal and boundary conditions. Results of tests will be documented in a test report.	Functional acceptance tests help to verify and validate the software.	All/All
2064	VRD-xxx-318	SAMS-PLN-002 section 10.7.4	Medium control level software efforts for SAMS shall undergo a Functional Configuration Audit (FCA) prior to the release of the software.	Medium control level software efforts for SAMS shall undergo a Functional Configuration Audit (FCA) prior to the release of the software.	Audits help to verify and validate the software.	All/All
2065	VRD-xxx-319	SAMS-PLN-002 section 10.7.4	Medium control level software efforts for SAMS shall undergo a Physical Configuration Audit (PCA) prior to the release of the software.	Medium control level software efforts for SAMS will undergo a Physical Configuration Audit (PCA) prior to the release of the software.	Audits help to verify and validate the software.	All/All
3001	See App. A	SSP 57000	SAMS shall meet ISS integrator requirements per Appendix A.	Reference Appendix A.	Necessary to meet carrier requirements in order to fly hardware.	ISS/VSS, AAS, QSS, CDHS

APPENDIX A – ISS REQUIREMENT APPLICABILITY MATRIX

The following table is derived from SSP 57000, Pressurized Payloads Interface Requirements Document, November 1, 2000, Revision E. For each requirement from SSP 57000, this table lists the requirement number, requirement title, the applicability, the applicable hardware, and comments. For applicability the choices are:

- A – Applicable
- F – Levied by the facility
- NAR – Not a requirement
- N/A – Not applicable

For hardware applicability the choices are:

- EE – Electronics Enclosure (60005MA12200 Rev F or equivalent)
- SE – Sensor Enclosure (60005MA12100 Rev E or equivalent)
- TSH-ES – Triaxial Sensor Head – Ethernet Standalone (60005MA50000 Rev A or equivalent)
- Cables – Cables between EE and SE or TSH-ES to facility

The hardware applicability category includes only subsystem hardware that is flying or planned to fly on ISS. All hardware that is considered a payload (ie ICU Drawer, RTS Drawer, MAMS, etc.) will be subject to the specific carrier requirements. If a design change occurs on the EE, SE or TSH-ES hardware/software, all requirements shall be reviewed for applicability.

The comments section lists only design specifications in SSP 57000 or lists SSP 57000 itself, as well as any other applicable information.

SSP 57000 Req. #	Verification #	Title	Applicability	Hardware Applicability				Comments
				EE	SE	TSH- ES	Cables	
1.0	N/A	Introduction	NAR	-	-	-	-	
1.1	N/A	Purpose	NAR	-	-	-	-	
1.2	N/A	Scope	NAR	-	-	-	-	
1.3	N/A	Use	NAR	-	-	-	-	
1.4	N/A	Exceptions	NAR	-	-	-	-	
1.5	N/A	Control and Maintenance	NAR	-	-	-	-	
2.0	N/A	Documentation	NAR	-	-	-	-	
2.1	N/A	Applicable Documents	NAR	-	-	-	-	
2.2	N/A	Reference Documents	NAR	-	-	-	-	
3.0	N/A	Payload Interface Requirements and Guidance	NAR	-	-	-	-	
3.1	N/A	Structural/Mechanical, and Microgravity and Stowage Interface Requirements	NAR	-	-	-	-	
3.1.1	N/A	Structural/Mechanical	NAR	-	-	-	-	
3.1.1.1 (A-D)	N/A	GSE Interfaces	N/A	-	-	-	-	SAMS is planning to interface to a rack/payload.
3.1.1.2 (A,E)	N/A	MPLM Interfaces	N/A	-	-	-	-	SAMS is planning to interface to a rack/payload.
3.1.1.2 (B)	VRD-xxx-250	MPLM Interfaces	A	X	X	X	-	Depress/repress rates, SSP 57000.
3.1.1.2 (C,D)	N/A	MPLM Interfaces	NAR	-	-	-	-	These requirements have been deleted.
3.1.1.2.1	N/A	MPLM Late/Early Access Requirements	N/A	-	-	-	-	SAMS has no plan to need late/early access.
3.1.1.2.1.1 (A-C)	N/A	MPLM Late Access Envelope (KSC)	N/A	-	-	-	-	SAMS has no plans to need late access.
3.1.1.2.1.2 (A,B)	N/A	MPLM Early Access Envelopes (KSC and DFRC)	N/A	-	-	-	-	SAMS has no plans to need early access.
3.1.1.3 (A,B,F)	VRD-xxx-201	Loads Requirements	F	X	X	X	-	A – SSP 41017 Part 1, para. 3.2.1.4.2, Part 2, para. 3.1.3, B – SSP 57000, F – SSP 57000, SSP 52005. Facility requirements shall be negotiated in the ICAD.
3.1.1.3 (C,E)	N/A	Loads Requirements	N/A	-	-	-	-	C – SAMS has no plans to have Rack Utility Panel umbilicals. E – SAMS has no plans to be mounted on ISPR posts.
3.1.1.3 (D)	VRD-xxx-251	Loads Requirements	A	X	X	X	-	Crew induced loads. SSP 57000.
3.1.1.4 (A,C,E,F,I,K,L, M)	N/A	Rack Requirements	N/A	-	-	-	-	SAMS has no plans to be an integrated rack.
3.1.1.4 (B)	VRD-xxx-207	Rack Requirements	F	X	X	X	-	On-orbit depress/repress rates. SSP 41002, par. 3.1.7.2.1. Facility requirements shall be negotiated in the ICAD.
3.1.1.4 (D)	VRD-xxx-205	Rack Requirements	A	X	X	X	-	Modal frequency, SSP 57000.
3.1.1.4 (G,H,J)	N/A	Rack Requirements	NAR	-	-	-	-	These requirements have been deleted.
3.1.1.4.1 (A-C)	N/A	Lab Window Rack Location Requirements	N/A	-	-	-	-	SAMS has no requirement for utilizing the lab window rack location.
3.1.1.5 (A)	VRD-xxx-201	Safety Critical Structures Requirements	A	X	X	X	-	SSP 52005.
3.1.1.5 (B-D)	N/A	Safety Critical Structures Requirements	NAR	-	-	-	-	These requirements have been deleted.

KEY: A – Applicable;

F – Levied by facility;

NAR – Not a requirement;

N/A – Not applicable

SSP 57000 Req. #	Verification #	Title	Applicability	Hardware Applicability				Comments
				EE	SE	TSH-ES	Cables	
3.1.1.6	N/A	Connector and Umbilical Physical Mate	NAR	-	-	-	-	
3.1.1.6.1	N/A	Connector Physical Mate	N/A	-	-	-	-	SAMS has no plans to mate to ISS directly.
3.1.1.6.2	N/A	Umbilical Physical Mate	N/A	-	-	-	-	SAMS has no plans to mate to ISS directly.
3.1.1.7 (A,B)	N/A	On-orbit Payload Protrusions	N/A	-	-	-	-	Protrusions are dependent on the mounting position for which the SAMS integrator will be responsible.
3.1.1.7.1	N/A	On-orbit Permanent Protrusions	N/A	-	-	-	-	Protrusions are dependent on the mounting position for which the SAMS integrator will be responsible.
3.1.1.7.2 (A-C)	N/A	On-orbit Semi-Permanent Protrusions	N/A	-	-	-	-	Protrusions are dependent on the mounting position for which the SAMS integrator will be responsible.
3.1.1.7.3 (A-B)	N/A	On-orbit Temporary Protrusions	N/A	-	-	-	-	Protrusions are dependent on the mounting position for which the SAMS integrator will be responsible.
3.1.1.7.4	N/A	On-orbit Momentary Protrusions	N/A	-	-	-	-	Protrusions are dependent on the mounting position for which the SAMS integrator will be responsible.
3.1.1.7.5 (A-B)	N/A	On-orbit Protrusions for Keep Alive Payloads	N/A	-	-	-	-	SAMS has no plans to be a keep alive payload.
3.1.2	N/A	Microgravity	NAR	-	-	-	-	
3.1.2.1 (A, B)	N/A	Quasi-steady Requirements	N/A	-	-	-	-	SAMS has no plans to have any disturbers.
3.1.2.2 (A,B)	N/A	Vibratory Requirements	N/A	-	-	-	-	SAMS has no plans to have any disturbers.
3.1.2.3 (A,B)	N/A	Transient Requirements	N/A	-	-	-	-	SAMS has no plans to have any disturbers.
3.1.2.4	N/A	Microgravity Environment	NAR	-	-	-	-	
3.1.2.5	N/A	ARIS Interfaces	NAR	-	-	-	-	
3.1.3	N/A	Stowage	NAR	-	-	-	-	
3.2	N/A	Electrical Interface Requirements	NAR	-	-	-	-	
3.2.1	N/A	Electrical Power Characteristics	NAR	-	-	-	-	
3.2.1.1	N/A	Steady-State Voltage Characteristics	NAR	-	-	-	-	
3.2.1.1.1	VRD-xxx-150	Interface B	F	X	-	X	-	SAMS is planning to interface to a rack with 28VDC or \pm 15VDC output. Facility requirements shall be negotiated in the ICAD.
3.2.1.1.2	VRD-xxx-150	Interface C	F	X	-	X	-	SAMS is planning to interface to a rack with 28VDC or \pm 15VDC output. Facility requirements shall be negotiated in the ICAD.
3.2.1.2		Ripple Voltage Characteristics	NAR	-	-	-	-	
3.2.1.2.1	VRD-xxx-102	Ripple Voltage and Noise	F	X	X	X	X	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.
3.2.1.2.2	VRD-xxx-102	Ripple Voltage Spectrum	F	X	X	X	X	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.
3.2.1.3	N/A	Transient Voltages	NAR	-	-	-	-	
3.2.1.3.1	VRD-xxx-102	Interface B	F	X	X	X	-	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.

KEY: A – Applicable;

F – Levied by facility;

NAR – Not a requirement;

N/A – Not applicable

SSP 57000 Req. #	Verification #	Title	Applicability	Hardware Applicability				Comments
				EE	SE	TSH-ES	Cables	
3.2.1.3.2	VRD-xxx-102	Interface C	F	X	X	X	-	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.
3.2.1.3.3	VRD-xxx-151	Fault Clearing and Protection	F	X	X	X	-	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.
3.2.1.3.4 (A,B)	VRD-xxx-150	Non-Normal Voltage Range	F	X	X	X	-	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.
3.2.1.4	N/A	Deleted	NAR	-	-	-	-	
3.2.2	N/A	Electrical Power Interface	NAR	-	-	-	-	
3.2.2.1 (A-F)	N/A	UIP and UOP Connectors and Pin Assignments	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.2.2.2 (A,B)	N/A	Power Bus Isolation	N/A	-	-	-	-	SAMS does not plan to require power from 2 buses.
3.2.2.3	VRD-xxx-152	Compatibility with Soft Start/Stop RPC	F	X	X	X	-	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.
3.2.2.4	VRD-xxx-152	Surge Current	F	X	X	X	-	SAMS is planning to interface to a rack. Performed as a system level test per verifications stated in AIDD, SAMS-INT-001.
3.2.2.5	N/A	Reverse Energy/Current	N/A	-	-	-	-	SAMS has no plans to have energy storage devices.
3.2.2.6	N/A	Circuit Protection Devices	NAR	-	-	-	-	
3.2.2.6.1	N/A	ISS EPS Circuit Protection Characteristics	NAR	-	-	-	-	
3.2.2.6.1.1 (A-E)	N/A	Remote Power Controllers (RPCS)	N/A	-	-	-	-	SAMS has no plans to have a RPCS.
3.2.2.6.2	N/A	EPCE RPC Interface Req.	NAR	-	-	-	-	
3.2.2.6.2.1	N/A	RPC Trip Coordination	NAR	-	-	-	-	
3.2.2.6.2.1.1	N/A	Payload Trip Ratings	N/A	-	-	-	-	SAMS does not plan to connect to Interface B.
3.2.2.6.2.1.2	N/A	Deleted	NAR	-	-	-	-	
3.2.2.7	N/A	EPCE Complex Load Impedances	NAR	-	-	-	-	
3.2.2.7.1 (A,B)	VRD-xxx-153	Interface B	F	X	X	X	-	Since SAMS has a specific impedance, it is up to the facility to incorporate SAMS into their design.
3.2.2.7.2	VRD-xxx-153	Interface C	F	X	X	X	-	Since SAMS has a specific impedance, it is up to the facility to incorporate SAMS into their design.
3.2.2.8	VRD-xxx-154	Large Signal Stability	F	X	X	X	-	SAMS has no plans to connect directly to interface B or C. Since SAMS hardware is low power, and voltage, it has no effect on the large signal stability. This requirement will be levied on the integrator per the AIDD, SAMS-INT-001.
3.2.2.9	VRD-xxx-102	Maximum Ripple Voltage Emissions	F	X	-	X	-	SAMS will characterize the system to verify hardware meets limits stated in AIDD, SAMS-INT-001.

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				EE	SE	TSH-ES	Cables	
3.2.2.10 (A-C)	VRD-xxx-102	Electrical Load-stand Alone Stability	F	X	X	X	-	SAMS will characterize the system to verify hardware meets limits stated in AIDD, SAMS-INT-001.
3.2.2.11	VRD-xxx-160	Electrical Load Inductance	F	X	X	X	-	SAMS will characterize the system to verify hardware meets limits stated in AIDD, SAMS-INT-001.
3.2.3	N/A	Electrical Power Consumer Constraints	NAR	-	-	-	-	
3.2.3.1(A,C)	N/A	Wire Derating	N/A	-	-	-	-	A – SAMS has no plans to have a connection between EPCE and the UOP. C - SAMS is planning to interface to a rack.
3.2.3.1 (B)	VRD-xxx-163	Wire Derating	A	X	X	-	X	SSP 57000, TM 102179 interpreted by NSTS 18798, TA-92-038.
3.2.3.2 (A,B)		Exclusive Power Feeds	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.2.3.3	VRD-xxx-155	Loss Of Power	A	X	X	X	-	NSTS 1700.7.
3.2.4	VRD-xxx-102	Electromagnetic Compatibility	F	X	X	X	X	SSP 30243, para. 3.1 & 3.6.2.
3.2.4.1	VRD-xxx-156	Electrical Grounding	A & F	X	X	X	-	SSP 30240, section 3. Any further requirements will be defined in the AIDD, SAMS-INT-001.
3.2.4.2	VRD-xxx-157	Electrical Bonding	A & F	X	X	X	X	SSP 30245 & NSTS 1700.7, sections 213 & 220. Any further requirements will be defined in the ICAD.
3.2.4.3	N/A	Cable/Wire Design and Control Requirements	N/A	-	-	-	-	SAMS does not plan to connect to Interface B or Interface C.
3.2.4.4	VRD-xxx-102	Electromagnetic Interference	A	X	X	X	X	SSP 30237, SSP 57000.
3.2.4.5	VRD-xxx-158	Electrostatic Discharge	A	X	X	X	-	SSP 57000.
3.2.4.6	VRD-xxx-102	Alternating Current (AC) Magnetic Fields	A	X	X	X	X	SSP 57000.
3.2.4.7	VRD-xxx-102	Direct Current (DC) Magnetic Fields	A	X	X	X	X	SSP 57000.
3.2.4.8	N/A	Corona	N/A	-	-	-	-	SAMS is planning to have low voltage.
3.2.4.9	N/A	Lightning	N/A	-	-	-	-	SAMS has no plans to power on during launch.
3.2.4.10	N/A	EMI Susceptibility for Safety-Critical Circuits	N/A	-	-	-	-	SAMS does not plan to have any safety-critical circuits.
3.2.5	N/A	Safety Requirements	NAR	-	-	-	-	
3.2.5.1	N/A	Payload Electrical Safety	NAR	-	-	-	-	
3.2.5.1.1	VRD-xxx-161	Mating/Demating of Powered Connectors	A	X	X	X	X	NSTS 1700.7 Addendum, NSTS 18798, MA2-97-093.
3.2.5.1.2	N/A	Safety-Critical Circuits Redundancy	N/A	-	-	-	-	SAMS does not plan to have any safety-critical circuits.
3.2.5.2	N/A	Rack Maintenance Switch (Rack Power Switch)	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.2.5.3 (A-C)	N/A	Power Switches/Controls	N/A	-	-	-	-	SAMS has no plans for voltage >30 Vrms.
3.2.5.4 (A-G)	N/A	Ground Fault Circuit Interrupters (GFCI)/Portable Equipment DC Sourcing Voltage	N/A	-	-	-	-	SAMS has no GFCI because SAMS has no plans for voltage >30 Vrms.
3.2.5.5 (A-B)	N/A	Portable Equipment/Power Cords	N/A	-	-	-	-	SAMS has no plans for portable equipment/power cords.
3.2.5.6	N/A	Deleted	NAR	-	-	-	-	
3.2.6	N/A	MPLM	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.1 (A-G)	N/A	MPLM Electrical Power Characteristics	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.1 (H,I)	N/A	MPLM Electrical Power Characteristics	NAR	-	-	-	-	These requirements have been deleted.

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3.2.6.2 (A-F)	N/A	MPLM Electrical Power Interface	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.2.1 (A-C)	N/A	MPLM UIP Connectors and Pin Assignments	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.2.2	N/A	Compatibility with RPC Soft Start/Stop in MPLM	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.2.3 (A-D)	N/A	MPLM Surge Current	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.2.4	N/A	MPLM Reverse Energy/Current	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.2.5	N/A	MPLM Payload Trip Ratings	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.3 (A-D)	N/A	MPLM Electrical Power Consumer Constraints	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.4 (A-J)	N/A	MPLM Electromagnetic Compatibility	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.4.1	N/A	MPLM Bonding	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.2.6.5 (A-F)	N/A	MPLM Safety Requirements	N/A	-	-	-	-	SAMS has no plans to use power in the MPLM.
3.3	N/A	Command and Data Handling Interface Requirements	NAR	-	-	-	-	
3.3.1	N/A	General Requirements	NAR	-	-	-	-	
3.3.2	VRD-xxx-tbd	Word/Byte Notations, Types and Data Transmissions	A	-	-	X	-	SSP 57000.
3.3.2.1	VRD-xxx-tbd	Word/Byte Notations	A	-	-	X	-	SSP 52050, para. 3.1.1.
3.3.2.2	VRD-xxx-tbd	Data Types	A	-	-	X	-	SSP 52050, para. 3.2.1.
3.3.2.3 (B)	VRD-xxx-tbd	Data Transmissions	A	-	-	X	-	SSP 52050, para. 3.3.3.1.
3.3.2.3 (A,C)	N/A	Data Transmissions	N/A	-	-	-	-	SAMS has no plans for using Medium Rate Data Link.
3.3.3	N/A	Deleted	NAR	-	-	-	-	
3.3.4	VRD-xxx-tbd	Consultative Committee for Space Data Systems	A	-	-	X	-	SSP 57000.
3.3.4.1 (A-C)	VRD-xxx-tbd	CCSDS Data	A	-	-	X	-	SSP 57000.
3.3.4.1.1	VRD-xxx-tbd	CCSDS Data Packets	A	-	-	X	-	SSP 52050 para. 3.1.3.
3.3.4.2.1.1	VRD-xxx-tbd	CCSDS Primary Header	A	-	-	X	-	SSP 52050 para. 3.1.3.1.
3.3.4.2.1.2 (A,B)	VRD-xxx-tbd	CCSDS Secondary Header	A	-	-	X	-	SSP 52050 para. 3.1.3.2, SSP 57000.
3.3.4.2.2	VRD-xxx-tbd	CCSDS Data Field	A	-	-	X	-	SSP 52050 para. 3.1
3.3.4.2.3	VRD-xxx-tbd	CCSDS Data Bitstream	A	-	-	X	-	701.0-B-2, para. 2.3.2.3.
3.3.4.2.4	VRD-xxx-tbd	CCSDS Application Process Identification Field	A	-	-	X	-	SSP 41175-2, para. 3.3.2.1.3 & Table 3.3.2.1.1-1.
3.3.4.3	N/A	CCSDS Time Codes	NAR	-	-	-	-	
3.3.4.3.1	VRD-xxx-tbd	CCSDS Unsegmented Time	A	-	-	X	-	CCSDS 301.0-B-2, para. 2.2.
3.3.4.3.2	VRD-xxx-tbd	CCSDS Segmented Time	A	-	-	X	-	SSP 57000, CCSDS 301.0-B-2.
3.3.5	N/A	MIL-STD-1553B Low Rate Data Link (LRDL)	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1	N/A	MIL-STD-1553B Protocol	NAR	-	-	-	-	
3.3.5.1.1	N/A	Standard Messages	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.2	N/A	Commanding	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.3	N/A	Health and Status Data	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.4 (A,B)	N/A	Safety Data	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.4.1	N/A	Caution and Warning	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.4.1.1	N/A	Class 1 – Emergency	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.4.1.2	N/A	Class 2 – Warning	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.4.1.3	N/A	Class 3 – Caution	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.4.1.4	N/A	Class 4 – Advisory	N/A	-	-	-	-	SAMS is not planning on using LRDL.

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3.3.5.1.5	N/A	Service Requests	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.6	N/A	Ancillary Data	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.7	N/A	File Transfer	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.8	N/A	Low Rate Telemetry	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.9	N/A	Defined Mode Codes	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.10	N/A	Implemented Mode Codes	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.11	N/A	Unimplemented/Undefined Mode Codes	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.1.12	N/A	Illegal Commands	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.2	N/A	MIL-STD-1553B Low Rate Data Link (LRDL) Interface Characteristics	NAR	-	-	-	-	
3.3.5.2.1	N/A	LRDL Remote Terminal Assignment	NAR	-	-	-	-	
3.3.5.2.1.1	N/A	LRDL Connector/Pin Assignments	NAR	-	-	-	-	
3.3.5.2.1.2 (A-C)	N/A	MIL-STD-1553B Bus A and B Connector/Pin Assignment	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.2.1.3	N/A	Deleted	NAR	-	-	-	-	
3.3.5.2.1.4 (A-E)	N/A	Remote Terminal Hardwired Address Coding	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.2.2	N/A	LRDL Signal Characteristics	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.2.3	N/A	LRDL Cabling	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.5.2.4	N/A	Multi-Bus Isolation	N/A	-	-	-	-	SAMS is not planning on using LRDL.
3.3.6	N/A	Medium Rate Data Link (MRDL)	NAR	-	-	-	-	
3.3.6.1	VRD-xxx-tbd	MRDL Protocol	A	-	-	X	-	SSP 52050, para. 3.3.
3.3.6.1.1	VRD-xxx-tbd	Integrated Rack Protocols on the MRDL	A	-	-	X	-	SSP 52050, para. 3.3, 3.3.4, 3.3.7.
3.3.6.1.2 (A-C)	VRD-xxx-tbd	MRDL Address	A	-	-	X	-	SSP 57000. B – Applies to TSH-ES that does not downlink data directly through ISS.
3.3.6.1.3 (A-C)	N/A	ISPR MRDL Connectivity	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.6.1.4 (A-D)	N/A	MRDL Connector/Pin Assignments and Wire Requirements	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.6.1.4.1	N/A	Deleted	NAR	-	-	-	-	
3.3.6.1.4.2	N/A	Deleted	NAR	-	-	-	-	
3.3.6.1.5	N/A	MRDL Signal Characteristics	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.6.1.6	N/A	MRDL Cable Characteristics	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.6.1.6.1	N/A	Insertion Loss	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.6.1.6.2	N/A	Differential Characteristic Impedance	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.6.1.6.3	N/A	Medium Timing Jitter	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.7	N/A	High Rate Data Link (HRDL)	NAR	-	-	-	-	
3.3.7.1	N/A	Payload to High Rate Frame Multiplexer (HRFM) Protocols	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.2	N/A	HRDL Interface Characteristics	NAR	-	-	-	-	
3.3.7.2.1	N/A	Physical Signaling	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.2.1.1 (A-C)	N/A	Physical Signaling Data Rates	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.2.2	N/A	Encoding	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.2.3	N/A	Deleted	NAR	-	-	-	-	
3.3.7.3	N/A	Integrated Rack HRDL Optical Power	NAR	-	-	-	-	

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3.3.7.3.1	N/A	Integrated Rack HRDL Transmitted Optical Power	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.3.2	N/A	Integrated Rack HRDL Received Optical Power	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.4	N/A	HRDL Fiber Optic Cable	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.5	N/A	HRDL Fiber Optic Cable Bend Radius	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.6 (A-D)	N/A	HRDL Connectors and Fiber	N/A	-	-	-	-	SAMS is not planning on using HRDL.
3.3.7.7	N/A	Deleted	NAR	-	-	-	-	
3.3.8	N/A	Personal Computers	NAR	-	-	-	-	
3.3.8.1 (A-H)	N/A	Payload Laptop	N/A	-	-	-	-	SAMS is not planning to provide a laptop to directly talk to a TSH-ES.
3.3.8.2 (A-C)	N/A	PCS	N/A	-	-	-	-	SAMS is not planning to provide a laptop to directly talk to a TSH-ES.
3.3.8.2.1 (A,B)	N/A	PCS to UOP Interface	N/A	-	-	-	-	SAMS is not planning to provide a laptop to directly talk to a TSH-ES.
3.3.8.2.2 (A-D)	N/A	760XD Laptop to Rack Interface	N/A	-	-	-	-	SAMS is not planning to provide a laptop to directly talk to a TSH-ES.
3.3.8.3 (A,B)	N/A	SSC	N/A	-	-	-	-	SAMS is not planning to use the SSC to directly talk to a TSH-ES.
3.3.9	N/A	UOP	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.10	N/A	Maintenance Switch, Smoke Detector, Smoke Indicator, and Integrated Rack Fan Interfaces	NAR	-	-	-	-	
3.3.10.1 (A,B)	N/A	Rack Maintenance Switch (Rack Power Switch) Interfaces	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.10.2	N/A	Smoke Detector Interfaces	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.10.2.1	N/A	Analog Interface Characteristics	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.10.2.2	N/A	Discrete Command Built-In-Test Interface Characteristics	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.10.2.3	N/A	Smoke Indicator Electrical Interfaces	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.10.2.4	N/A	Fan Ventilation Status Electrical Interfaces	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.3.10.3 (A-C)	N/A	Rack Maintenance Switch (Rack Power Switch)/Fire Detection Support Interface Connector	N/A	-	-	-	-	Provided by integrator as per AIDD.
3.4	N/A	Payload NTSC Video and Audio Interface Requirements	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1	N/A	Payload NTSC Video Interface Requirements	NAR	-	-	-	-	
3.4.1.1 (A-C)	N/A	Payload NTSC Optical Video Characteristics	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.2	N/A	NTSC Fiber Optic Video	NAR	-	-	-	-	
3.4.1.2.1 (A,B)	N/A	Pulse Frequency Modulation NTSC Fiber Optic Video Characteristics	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.2.2	N/A	Integrated Rack NTSC PFM Video Transmitted Optical Power	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.2.3	N/A	Integrated Rack NTSC PFM Video and Sync Signal Received Optical Power	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.2.4	N/A	Fiber Optic Cable Characteristics	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.

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3.4.1.2.5	N/A	PFM NTSC Video Fiber Optic Cable Bend Radius	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.2.6	N/A	Deleted	NAR	-	-	-	-	
3.4.1.2.7 (A-C)	N/A	PFM NTSC Optical Connector/Pin Assignments	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.3	N/A	NTSC Electrical Video Interfaces	NAR	-	-	-	-	
3.4.1.3.1	N/A	Cables	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.3.2	N/A	Signal Standard	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.3.3	N/A	Interface Circuit	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.3.4	N/A	Cross Talk	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.1.4 (A-C)	N/A	NTSC Electrical Connector/Pin Assignments	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.4.2	N/A	U.S. Element Audio Interface Requirements	N/A	-	-	-	-	SAMS has no plans to use NTSC Video and Audio.
3.5	N/A	Thermal Control Interface Requirements	NAR	-	-	-	-	
3.5.1	N/A	Internal Thermal Control System (ITCS) Interface Requirements	NAR	-	-	-	-	
3.5.1.1 (A,B)	N/A	Physical Interface	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.2 (A,B)	N/A	ITCS Fluid Use and Charging	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.3	N/A	ITCS Pressure Drop	NAR	-	-	-	-	
3.5.1.3.1 (A,B)	N/A	On-Orbit Interfaces	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.3.2	N/A	MPLM Interfaces	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.4 (A,B)	N/A	Coolant Flow Rate	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.5 (A,B)	N/A	Coolant Supply Temperature	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.6 (A-D)	N/A	Coolant Return Temperature	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.7 (A-C)	N/A	Coolant Maximum Design Pressure	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.8	N/A	Fail Safe Design	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.9 (A,B)	N/A	Leakage	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.10	N/A	Quick-Disconnect Air Inclusion	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.11	N/A	Rack Front Surface Temperature	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.12	N/A	Cabin Air Heat Leak	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.13	N/A	MPLM Cabin Air Cooling	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.14 (A,B)	N/A	Simultaneous Cooling	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.15	N/A	Control System Time Constant	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.16	N/A	Payload Coolant Quantity	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.5.1.17	N/A	Payload Gas Inclusion (TBR #6)	N/A	-	-	-	-	SAMS is not planning to be actively cooled.
3.6	N/A	Vacuum System Requirements	NAR	-	-	-	-	
3.6.1	N/A	Vacuum Exhaust System (VES)/Waste Gas System (WGS) Requirements	NAR	-	-	-	-	
3.6.1.1	N/A	VES/WGS Physical Interface	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.1.2 (A-C)	N/A	Input Pressure Limit	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.1.3	N/A	Input Temperature Limit	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.

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3.6.1.4	N/A	Input Dewpoint Limit	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.1.5 (A-D)	N/A	Acceptable Exhaust Gases	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.1.5.1 (A-C)	N/A	Acceptable Gases – List	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.1.5.2	N/A	External Contamination Control	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.1.5.3 (A,B)	N/A	Incompatible Gases	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.1.6	N/A	Payload Vacuum System Access Valve	N/A	-	-	-	-	SAMS is not planning to need the VES or WGS.
3.6.2	N/A	Vacuum Resource System (VRS)/Vacuum Vent System (VVS) Requirements	NAR	-	-	-	-	
3.6.2.1	N/A	VRS/VVS Physical Interface	N/A	-	-	-	-	SAMS is not planning to need the VRS or VVS.
3.6.2.2 (A-C)	N/A	Input Pressure Limit	N/A	-	-	-	-	SAMS is not planning to need the VRS or VVS.
3.6.2.3	N/A	VRS/VVS Through-put Limit	N/A	-	-	-	-	SAMS is not planning to need the VRS or VVS.
3.6.2.4	N/A	Acceptable Gases	N/A	-	-	-	-	SAMS is not planning to need the VRS or VVS.
3.7	N/A	Pressurized Gases Interface Requirement	NAR	-	-	-	-	
3.7.1	N/A	Nitrogen Interface Requirements	NAR	-	-	-	-	
3.7.1.1	N/A	Nitrogen Interface Control	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.1.2	N/A	Nitrogen Interface MDP	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.1.3	N/A	Nitrogen Interface Temperature	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.1.4	N/A	Nitrogen Leakage	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.1.5	N/A	Nitrogen Physical Interface	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.2	N/A	Argon Interface Requirements	NAR	-	-	-	-	
3.7.2.1	N/A	Argon Interface Control	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.2.2	N/A	Argon Interface MDP	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.2.3	N/A	Argon Interface Temperature	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.2.4	N/A	Argon Leakage	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.2.5	N/A	Argon Physical Interface	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.3	N/A	Carbon Dioxide Interface Requirements	NAR	-	-	-	-	
3.7.3.1	N/A	Carbon Dioxide Interface Control	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.3.2	N/A	Carbon Dioxide Interface MDP	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.3.3	N/A	Carbon Dioxide Interface Temperature	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.3.4	N/A	Carbon Dioxide Interface Leakage	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.3.5	N/A	Carbon Dioxide Physical Interface	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.4	N/A	Helium Interface Requirements	NAR	-	-	-	-	
3.7.4.1	N/A	Helium Interface Control	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.4.2	N/A	Helium Interface MDP	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.4.3	N/A	Helium Interface Temperature	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.4.4	N/A	Helium Leakage	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.4.5	N/A	Helium Physical Interface	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.5	N/A	Pressurized Gas Systems	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.7.6	N/A	Manual Valves	N/A	-	-	-	-	SAMS has no plans to use pressurized gases.
3.8	N/A	Payload Support Services Interfaces Requirements	NAR	-	-	-	-	
3.8.1	N/A	Potable Water	NAR	-	-	-	-	
3.8.1.1	N/A	Potable Water Interface Connection	N/A	-	-	-	-	SAMS does not plan to use any payload support services.
3.8.1.2	N/A	Potable Water Interface Pressure	N/A	-	-	-	-	SAMS does not plan to use any payload support services.

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3.8.1.3 (A,B)	N/A	Potable Water Use	N/A	-	-	-	-	SAMS does not plan to use any payload support services.
3.8.2	N/A	Fluid System Servicer	N/A	-	-	-	-	SAMS does not plan to use any payload support services.
3.9	N/A	Environment Interface Requirements	NAR	-	-	-	-	
3.9.1	N/A	Atmosphere Requirements	NAR	-	-	-	-	
3.9.1.1	VRD-xxx-252	Pressure	A	X	X	X	-	SSP 57000.
3.9.1.2	VRD-xxx-209	Temperature	A	X	X	X	X	SSP 57000.
3.9.1.3	VRD-xxx-209	Humidity	A	X	X	X	X	SSP 57000.
3.9.2	N/A	Integrated Rack Use of Cabin Atmosphere	NAR	-	-	-	-	
3.9.2.1 (A,B)	N/A	Active Air Exchange	N/A	-	-	-	-	SAMS does not plan to need active air exchange.
3.9.2.2	N/A	Oxygen Consumption	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.9.2.3	N/A	Chemical Releases	N/A	-	-	-	-	SAMS does not plan to release chemicals to the cabin air.
3.9.3	N/A	Radiation Requirements	NAR	-	-	-	-	
3.9.3.1	N/A	Integrated Rack Contained or Generated Ionizing Radiation	N/A	-	-	-	-	SAMS does not plan on containing or generating ionizing radiation.
3.9.3.2	N/A	Ionizing Radiation Dose	N/A	-	-	-	-	The shielding of any ISS module is adequate to absorb the ionizing radiation dose.
3.9.3.3	VRD-xxx-104	Single Event Effect (SEE) Ionizing Radiation	A	X	X	X	-	SSP 57000.
3.9.3.4	N/A	Lab Window Rack Location Radiation Requirements	N/A	-	-	-	-	SAMS does not plan to remove the protective shield.
3.9.3.4.1	N/A	Window Rack Infrared Radiation Requirements	N/A	-	-	-	-	SAMS does not plan to remove the protective shield.
3.9.3.4.2	N/A	Window Rack Ultraviolet Radiation Requirements	N/A	-	-	-	-	SAMS does not plan to remove the protective shield.
3.9.4	VRD-xxx-253	Additional Environmental Conditions	A	X	X	X	-	SSP 57000. SAMS will verify pressures from 13.0 psia – 15.0 psia and humidity from 20% to 80%.
3.10	N/A	Fire Protection Interface Requirements	NAR	-	-	-	-	
3.10.1	VRD-xxx-451	Fire Prevention	A	X	X	X	-	NSTS 1700.7, Addendum, par. 220.10a. The cable is part of the system, alone this requirement is N/A for cables.
3.10.2	N/A	Payload Monitoring and Detection Requirements	NAR	-	-	-	-	
3.10.2.1	N/A	Smoke Detection	NAR	-	-	-	-	
3.10.2.1.1 (A,B)	VRD-xxx-451	Smoke Detector	F	X	X	X	-	SAMS does not plan to have a smoke detector. The integrator will be responsible for this requirement dependent on the PSRP review.
3.10.2.1.2	N/A	Forced Air Circulation Indication	N/A	-	-	-	-	SAMS does not plan on having forced air circulation. The integrator will be responsible for this requirement dependent on the PSRP review.
3.10.2.1.3 (A,B)	N/A	Fire Detection Indicator	N/A	-	-	-	-	SAMS does not plan to have a fire detection indicator. The integrator will be responsible for this requirement dependent on the PSRP review.
3.10.2.2	N/A	Parameter Monitoring (TBR #10)	NAR	-	-	-	-	

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3.10.2.2.1	VRD-xxx-451	Parameter Monitoring Use	A	X	X	X	-	SSP 57000. This is dependent on the PSRP review.
3.10.2.2.2	N/A	Parameter Monitoring in Response	NAR	-	-	-	-	
3.10.2.2.2.1 (A,B)	N/A	Parameter Monitoring in Subrack	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.10.2.2.2.2 (A,B)	N/A	Parameter Monitoring in Integrated Rack	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.10.3	N/A	Fire Suppression	NAR	-	-	-	-	
3.10.3.1 (A,B)	VRD-xxx-451	Portable Fire Extinguisher	F	X	X	X	-	The requirement of access ports will be dependent on the PSRP review.
3.10.3.2	VRD-xxx-451	Fire Suppression Access Port Accessibility	F	X	X	X	-	The requirement of access ports will be dependent on the PSRP review.
3.10.3.3	N/A	Fire Suppressant Distribution	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.10.3.4	N/A	Deleted	NAR	-	-	-	-	
3.10.3.4.1	N/A	Deleted	NAR	-	-	-	-	
3.10.3.4.2	N/A	Deleted	NAR	-	-	-	-	
3.10.4 (A,B)	VRD-xxx-451	Labeling	F	X	X	X	-	The requirement of labeling the access ports will be dependent on the PSRP review.
3.11	N/A	Materials and Parts Interface Requirements	NAR	-	-	-	-	
3.11.1	VRD-xxx-254	Materials and Parts Use and Selection	A	X	X	X	X	NSTS 1700.7, ISS Addendum.
3.11.1.1	VRD-xxx-254	Commercial Parts	A	X	X	X	X	NSTS 1700.7, ISS Addendum.
3.11.2 (A-C)	N/A	Fluids	N/A	-	-	-	-	SAMS does not plan to have any fluids.
3.11.3	VRD-xxx-219	Cleanliness	A	X	X	X	X	Visibly Clean – Sensitive as per SN-C-0005.
3.11.4	VRD-xxx-216	Fungus Resistant Material	A	X	X	X	X	SSP 30233, par. 4.2.10.
3.12	N/A	Human Factors Interface Requirements	NAR	-	-	-	-	
3.12.1 (A,B)	VRD-xxx-255	Strength Requirements	A	X	X	X	X	SSP 57000.
3.12.2	N/A	Body Envelope and Reach Accessibility	NAR	-	-	-	-	
3.12.2.1	N/A	Adequate Clearance	N/A	-	-	-	-	Clearance is dependent on a mounting position for which the SAMS integrator will be responsible.
3.12.2.2 (A,B)	VRD-xxx-256	Accessibility	A	X	X	X	X	SSP 57000. SAMS is designed for accessibility. The integrator is responsible for the accessibility in a specific mounted location.
3.12.2.3	VRD-xxx-256	Full Size Range Accommodation	A	X	X	X	X	SSP 57000. SAMS is designed for full size range accommodation. The integrator is responsible for the full size range accommodation in a specific mounted location.
3.12.3	N/A	Habitability	NAR	-	-	-	-	
3.12.3.1	N/A	Housekeeping	NAR	-	-	-	-	
3.12.3.1.1	VRD-xxx-tbd	Closures or Covers	A	X	X	-	-	SSP 57000. No covers needed for TSH-ES and cables.
3.12.3.1.2 (A,B)	N/A	Built-in Control	N/A	-	-	-	-	A – SAMS does not have plans to contain any liquids or particulate matter. B – SAMS does not have plans to incorporate a capture element.
3.12.3.1.2 (C)	N/A	Built-in Control	NAR	-	-	-	-	This requirement has been deleted.
3.12.3.1.3	N/A	Deleted	NAR	-	-	-	-	
3.12.3.1.4	N/A	Deleted	NAR	-	-	-	-	

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3.12.3.1.5	N/A	One-handed Operation	N/A	-	-	-	-	SAMS has no plans for cleaning equipment.
3.12.3.1.6	N/A	Deleted	NAR	-	-	-	-	
3.12.3.2	N/A	Touch Temperature	NAR	-	-	-	-	
3.12.3.2.1	VRD-xxx-210	Continuous/Incidental Contact-High Temperature	A	X	X	X	-	SSP 57000, NASA interpretation letter JSC, MA2-95-048.
3.12.3.2.2	N/A	Continuous/Incidental Contact-Low Temperature	N/A	-	-	-	-	SAMS has no plans for a refrigeration system.
3.12.3.3	N/A	Acoustic Requirements	N/A	-	-	-	-	SAMS has no plans for a noise source.
3.12.3.3.1 (A-C)	N/A	Continuous Noise Limits	N/A	-	-	-	-	SAMS has no plans for a noise source.
3.12.3.3.2	N/A	Intermittent Noise Limits	N/A	-	-	-	-	SAMS has no plans for a noise source.
3.12.3.4 (A)	N/A	Lighting Design	N/A	-	-	-	-	The EE is not in an exposed area. The SE and TSH-ES are small and can be located in various places. The traditional color of a SAMS sensor head is gold.
3.12.3.4 (B-E)	N/A	Lighting Design	N/A	-	-	-	-	The SAMS integrator is responsible for lighting conditions.
3.12.4	N/A	Structural/Mechanical Interfaces	NAR	-	-	-	-	
3.12.4.1	N/A	Deleted	NAR	-	-	-	-	
3.12.4.1.1	N/A	Deleted	NAR	-	-	-	-	
3.12.4.2	N/A	Payload Hardware Mounting	NAR	-	-	-	-	
3.12.4.2.1	VRD-xxx-256	Equipment Mounting	A	X	X	X	X	SSP 57000.
3.12.4.2.2	N/A	Drawers and Hinged Panels	N/A	-	-	-	-	SAMS does not plan to use drawers or hinged panels.
3.12.4.2.3	N/A	Deleted	NAR	-	-	-	-	
3.12.4.2.4	N/A	Deleted	NAR	-	-	-	-	
3.12.4.2.5	N/A	Alignment	N/A	-	-	-	-	SAMS does not plan to use blind mate connectors.
3.12.4.2.5.1	N/A	Deleted	NAR	-	-	-	-	
3.12.4.2.6	N/A	Slide-out Stops	N/A	-	-	-	-	SAMS does not plan to use slide or pivot mounted hardware.
3.12.4.2.7	N/A	Push-pull Force	N/A	-	-	-	-	SAMS does not plan to require a push-pull action.
3.12.4.2.8	N/A	Access	N/A	-	-	-	-	SAMS does not plan to be accessed on a daily/weekly basis.
3.12.4.2.8.1 (A,B)	N/A	Covers	N/A	-	-	-	-	SAMS does not plan to need any covers.
3.12.4.2.8.2	N/A	Self-supporting Covers	N/A	-	-	-	-	SAMS does not plan to have covers.
3.12.4.2.8.3	N/A	Deleted	NAR	-	-	-	-	
3.12.4.2.8.4	N/A	Unique Tools	N/A	-	-	-	-	SAMS does not plan to use any unique tools.
3.12.4.3	N/A	Connectors	NAR	-	-	-	-	
3.12.4.3.1	VRD-xxx-161	One-Handed Operation	A	X	X	X	-	SSP 57000. The SAMS integrator is responsible for the accessibility of the SAMS unit in its mounted location.
3.12.4.3.2 (A,B)	VRD-xxx-161	Accessibility	A	X	X	X	X	SSP 57000. The SAMS integrator is responsible for the accessibility of the SAMS unit in its mounted location.

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3.12.4.3.3 (A,B)	VRD-xxx-161	Ease of Disconnect	A	X	X	X	X	SSP 57000. For cables connecting to the SAMS integrator, SAMS will not inspect the integrator provided connector.
3.12.4.3.4	N/A	Indication of Pressure/Flow	N/A	-	-	-	-	SAMS has no planned fluid or gas lines.
3.12.4.3.5	VRD-xxx-161	Self Locking	A	X	X	X	X	SSP 57000.
3.12.4.3.6 (A,B)	VRD-xxx-161	Connector Arrangement	A	X	X	X	-	SSP 57000. The SAMS integrator is responsible for the arrangement of the SAMS designated connector(s) mounted on their hardware.
3.12.4.3.7	VRD-xxx-161	Arc Containment	A	X	X	X	X	SSP 57000.
3.12.4.3.8	VRD-xxx-161	Connector Protection	A	X	X	X	X	SSP 57000.
3.12.4.3.9	VRD-xxx-161	Connector Shape	A	X	X	X	X	SSP 57000.
3.12.4.3.10	N/A	Fluid and Gas Line Connectors	N/A	-	-	-	-	SAMS has no planned fluid or gas lines.
3.12.4.3.11 (A)	VRD-xxx-161	Alignment Marks or Guide Pins	A	X	X	X	X	SSP 57000.
3.12.4.3.11 (B)	N/A	Alignment Marks or Guide Pins	NAR	-	-	-	-	This requirement has been deleted.
3.12.4.3.12 (A,B)	VRD-xxx-161	Coding	A	X	X	X	X	SSP 57000.
3.12.4.3.13	VRD-xxx-161	Pin Identification	A	X	X	X	X	SSP 57000.
3.12.4.3.14	VRD-xxx-161	Orientation	A	X	X	X	X	SSP 57000.
3.12.4.3.15	N/A	Hose/Cable Restraints	N/A	-	-	-	-	The routing and restraining of SAMS cables is dependent on the mounting position of the SAMS unit for which the SAMS integrator will be responsible.
3.12.4.4	N/A	Fasteners	NAR	-	-	-	-	
3.12.4.4.1	N/A	Non-Threaded Fasteners Status Indication	N/A	-	-	-	-	SAMS has no plans for non-threaded fasteners.
3.12.4.4.2	VRD-xxx-257	Mounting Bolt/Fastener Spacing	A	X	X	X	-	SSP 57000.
3.12.4.4.3	N/A	Deleted	NAR	-	-	-	-	
3.12.4.4.4 (A)	VRD-xxx-257	Multiple Fasteners	A	X	X	X	-	SSP 57000. This requirement applies to mounting fasteners only.
3.12.4.4.4 (B)	N/A	Multiple Fasteners	NAR	-	-	-	-	This requirement has been deleted.
3.12.4.4.5	VRD-xxx-257	Captive Fasteners	A	X	X	X	-	SSP 57000.
3.12.4.4.6 (A, B)	N/A	Quick Release Fasteners	N/A	-	-	-	-	SAMS has no plans to use quick release fasteners.
3.12.4.4.7	VRD-xxx-257	Threaded Fasteners	A	X	X	X	-	SSP 57000.
3.12.4.4.8 (A- C)	N/A	Over Center Latches	N/A	-	-	-	-	SAMS has no plans to use latches.
3.12.4.4.9	N/A	Winghead Fasteners	N/A	-	-	-	-	SAMS has no plans to use winghead fasteners.
3.12.4.4.10	N/A	Deleted	NAR	-	-	-	-	
3.12.4.4.11 (A- C)	VRD-xxx-257	Fastener Head Type	A	X	X	X	-	SSP 57000.
3.12.4.4.12	VRD-xxx-257	One-Handed Actuation	A	X	X	X	-	SSP 57000.
3.12.4.4.13	N/A	Deleted	NAR	-	-	-	-	
3.12.4.4.14	N/A	Access Holes	N/A	-	-	-	-	SAMS has no plans for covers or shields. All SAMS mounting fasteners are in plain sight.
3.12.5	N/A	Controls and Displays	NAR	-	-	-	-	
3.12.5.1	VRD-xxx-tbd	Controls Spacing Design Requirements	A	X	-	-	-	SSP 57000. Applies to SAMS hardware that contains switches.

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3.12.5.2	VRD-xxx-tbd	Accidental Actuation	A	X	-	-	-	SSP 57000. Applies to SAMS hardware that contains switches.
3.12.5.2.1 (A-G)	VRD-xxx-tbd	Protective Methods	A	X	-	-	-	SSP 57000. Applies to SAMS hardware that contains switches.
3.12.5.2.2	VRD-xxx-tbd	Noninterference	A	X	-	-	-	SSP 57000. Applies to SAMS hardware that contains switches.
3.12.5.2.3	N/A	Dead-Man Controls	N/A	-	-	-	-	SAMS is a benign payload and does not require dead-man controls.
3.12.5.2.4	VRD-xxx-tbd	Barrier Guards	A	X	-	-	-	SSP 57000. Applies to SAMS hardware that contains switches.
3.12.5.2.5	N/A	Recessed Switch Protection	N/A	-	-	-	-	SAMS has no plans for rotary switches.
3.12.5.2.6	N/A	Deleted	NAR	-	-	-	-	
3.12.5.2.7	N/A	Position Indication	N/A	-	-	-	-	SAMS has no plans for switch protective covers.
3.12.5.2.8	N/A	Hidden Controls	N/A	-	-	-	-	SAMS has no plans for hidden controls.
3.12.5.2.9	N/A	Hand Controllers	N/A	-	-	-	-	SAMS has no plans for hand controllers.
3.12.5.3 (A-E)	N/A	Valve Controls	N/A	-	-	-	-	SAMS has no plans for valves.
3.12.5.4	VRD-xxx-tbd	Toggle Switches	A	X	-	-	-	SSP 57000.
3.12.6	N/A	Restraints and Mobility Aids	N/A	-	-	-	-	SAMS is planning to interface to a rack.
3.12.6.1 (A-C)	N/A	Stowage Drawer Contents Restraints	N/A	-	-	-	-	SAMS does not have plans for a stowage drawer.
3.12.6.2 (A,B)	N/A	Stowage and Equipment Drawers/Trays	N/A	-	-	-	-	SAMS does not have plans for stowage and equipment drawers/trays.
3.12.6.3	VRD-xxx-452	Captive Parts	A	X	X	X	X	SSP 57000.
3.12.6.4	N/A	Handle and Grasp Area Design Requirements	NAR	-	-	-	-	
3.12.6.4.1	N/A	Handles and Restraints	N/A	-	-	-	-	All SAMS hardware, except the EE, is planning to be grasped with one hand. The integrator of the EE is responsible, if it chooses to move an EE during nominal operations.
3.12.6.4.2	N/A	Deleted	NAR	-	-	-	-	
3.12.6.4.3	N/A	Handle Location/Front Access	N/A	-	-	-	-	All SAMS hardware, except the EE, is planning to be grasped with one hand. The integrator of the EE is responsible, if it chooses to move an EE during nominal operations.
3.12.6.4.4	N/A	Handle Dimensions	N/A	-	-	-	-	All SAMS hardware, except the EE, is planning to be grasped with one hand. The integrator of the EE is responsible, if it chooses to move an EE during nominal operations.
3.12.6.4.5 (A-C)	N/A	Non-Fixed Handles Design Requirements	N/A	-	-	-	-	SAMS has no plans for non-fixed handles.
3.12.7	VRD-xxx-453	Identification Labeling	A	X	X	X	X	SSP 57000, Appendix C.
3.12.7.1	N/A	Deleted	NAR	-	-	-	-	
3.12.7.2	N/A	Deleted	NAR	-	-	-	-	
3.12.7.3	N/A	Deleted	NAR	-	-	-	-	
3.12.7.4	N/A	Deleted	NAR	-	-	-	-	
3.12.7.5	N/A	Deleted	NAR	-	-	-	-	
3.12.7.6	N/A	Deleted	NAR	-	-	-	-	

KEY: A – Applicable;

F – Levied by facility;

NAR – Not a requirement;

N/A – Not applicable

SSP 57000 Req. #	Verification #	Title	Applicability	Hardware Applicability				Comments
				EE	SE	TSH-ES	Cables	
3.12.8	N/A	Color	N/A	-	-	-	-	SAMS color will be defined in the AIDD, SAMS-INT-001.
3.12.9	N/A	Crew Safety	NAR	-	-	-	-	
3.12.9.1 (A-E)	VRD-xxx-162	Electrical Hazards	A	X	X	X	-	SSP 57000. The cable is part of the system, alone this requirement is N/A for cables.
3.12.9.1.1	VRD-xxx-161	Mismatched	A	X	X	X	X	SSP 57000.
3.12.9.1.2	N/A	Deleted	NAR	-	-	-	-	
3.12.9.1.3	N/A	Deleted	NAR	-	-	-	-	
3.12.9.1.4	N/A	Overload Protection	NAR	-	-	-	-	
3.12.9.1.4.1	VRD-xxx-tbd	Device Accessibility	A	X	-	-	-	SSP 57000.
3.12.9.1.4.2	N/A	Extractor-Type Fuse Holder	N/A	-	-	-	-	SAMS has no plans to replace fuses on orbit.
3.12.9.1.4.3	VRD-xxx-tbd	Overload Protection Location	A	X	-	-	-	SSP 57000.
3.12.9.1.4.4	VRD-xxx-tbd	Overload Protection Identification	A	X	-	-	-	SSP 57000.
3.12.9.1.4.5	VRD-xxx-tbd	Automatic Restart Protection	A	X	-	-	-	SSP 57000.
3.12.9.1.5	N/A	Deleted	NAR	-	-	-	-	
3.12.9.1.5.1	N/A	Deleted	NAR	-	-	-	-	
3.12.9.2	VRD-xxx-258	Sharp Edges and Corners Protection	A	X	X	X	X	NSTS 1700.7, ISS Addendum, paragraph 222.1.
3.12.9.3	VRD-xxx-258	Holes	A	X	X	X	-	SSP 57000.
3.12.9.4	N/A	Latches	N/A	-	-	-	-	SAMS has no plans for latches.
3.12.9.5	VRD-xxx-258	Screws and Bolts	A	X	X	-	X	SSP 57000.
3.12.9.6	N/A	Securing Pins	N/A	-	-	-	-	SAMS has no plans for securing pins.
3.12.9.7	VRD-xxx-tbd	Levers, Cranks, Hooks, and Controls	A	X	-	-	-	SSP 57000.
3.12.9.8	VRD-xxx-258	Burrs	A	X	X	X	X	SSP 57000.
3.12.9.9 (A,B)	VRD-xxx-257	Locking Wires	A	X	X	X	-	SSP 57000, SSP 52005 para. 5.6.
3.12.9.10 (A, C,D)	N/A	Audio Devices (Displays)	N/A	-	-	-	-	SAMS has no plans for audio devices.
3.12.9.10 (B)	N/A	Audio Devices (Displays)	NAR	-	-	-	-	This requirement has been deleted.
3.12.9.11	N/A	Deleted	NAR	-	-	-	-	
3.12.9.12	N/A	Egress	N/A	-	-	-	-	This is a responsibility of the SAMS integrator as per the ICAD.
3.12.10	VRD-xxx-454	Payload In-Flight Maintenance	A	X	X	X	-	SSP 57000.
3.12.11	N/A	Deleted	NAR	-	-	-	-	

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NAR – Not a requirement;

N/A – Not applicable